

# ENGINEERING REPORT

REPORT SERIAL NO. 1720

DATE 2-4-75 PAGE 1 OF 8

Electrical Components Division

The Bendix Corporation

Sidney, New York

## NAME OF EQUIPMENT

INTERFACIAL INSERT FOR ELECTRICAL CONNECTORS

## TYPE &amp; PART NO.

## SERIAL NO.

## CUSTOMER'S NAME &amp; ADDRESS

Lyndon B. Johnson Space Center, NASA  
R & T Procurement Branch  
Houston, Texas 77058

NASA CR-

141728

## CUSTOMER'S PART NO.

TERMINATION REPORT  
FOR PROJECT 70886NAS 9-13899

## PURPOSE

Provide connector assemblies and perform design evaluation as required by NASA Contract NAS9-13899 and Bendix Technical Proposal 1003.

## SUBJECT

(NASA-CR-141728) INTERFACIAL INSERT FOR  
ELECTRICAL CONNECTORS Engineering Report  
(Bendix Corp.) 22 p HC \$3.25 CACL 09A

## 1. Connector Assemblies

- 1.1 25 interfacial insert assembly kits (L-25274-603) have been shipped per contract requirements. Each assembly consists of one interfacial insert with an interfacial seal on one side, pin contacts molded in place, two mating socket insert assemblies and 42 socket contacts for assembly into socket inserts.

The interfacial insert and ancillary components were designed to be assembled into modified MIL-C-26482 series II connectors to be provided on a possible follow up contract.

N75-19537

## 2. Design Evaluation

## 2.1 Conclusions

Unclas  
G3/33 14452

- 2.1.1 The samples and testing performed on this program indicated that the interfacial insert L-25274-604 can be assembled into specially modified MIL-C-26482 series II connectors.
- 2.1.2 The interfacial insert concept can be utilized in many other standard series connectors.
- 2.1.3 One electrical connector, together with an assembly tool not required by this contract, has been modified and built to accept an interfacial insert. This connector will be sent to Bob Hendrix, Technical Monitor. The connector will mate with any MIL-C-26482 plug assembly having the 22-21S arrangement in the normal position.

## 2.2 Design Comments for Future Development

- 2.2.1 Interfacial inserts manufactured for this contract were made from standard hard epoxy molds. New molds should be made to provide a contour more suitable for insertion and removal of the interfacial insert. The molding gate should be reduced in size so it can be trimmed below the sealing surface.

PREPARED	<i>D. P. H. H. H.</i>	APPROVED	<i>D. H. H. H.</i>	APPROVED		APPROVED	
DATE	2-6-75	DATE	2-6-75	DATE		DATE	

- 2.2.2 A more reliable insert-to-seal bond must be incorporated. The one used on these samples did not survive the temperature life testing.
- 2.2.3 The interfacial insert L-25274-604 is visually aligned and positioned in the shell. An assembly tool is used to seat the interfacial insert in the shell. A visual check is required to assure that the interfacial insert is properly seated.
- 2.2.4 Future testing of interfacial inserts made from new molds, assembled in modified shells and mated with standard connectors of the type desired to be qualified will be required to assure the design's performance.
- 2.2.5 During testing, particular attention should be given to the circumferential seal after exposure to the environments for excess wear of contacts and maintenance of insert position during and after all testing.
- 2.2.6 For connectors not having asymmetrical insert patterns with respect to the main key or keyway, the standard socket insert can not be used in the interfacial insert connector.
- 2.2.7 See L-25274-604 & L-25274-602 for illustration and modification dimensions for interfacial insert, modified plug and receptacle connector assemblies.

### 2.3 Study Requirements

- 2.3.1 Upon receipt of NASA's request to quote on the interfacial insert for electrical connectors, a review of the concept was undertaken. Various methods of inserting the interfacial insert and retaining it in a modified shell were considered. Methods investigated included separate inserts with an extra threaded retaining member or locking springs.
- It was decided that the best way to meet the design requirements and provide an economical end unit, the locking mechanism and interfacial insert should be one unit. Bendix proposal No. 1003 "Interfacial Inserts for Electrical Connectors" was submitted and accepted by NASA for prototype assembly.
- The approach chosen by Bendix offers the following design features:
- Retention device is molded as part of interfacial insert
  - Interfacial insert is molded using silicone rubber - different rubber materials can be used to accommodate special environmental situations such as unusual temperature extremes or unusual fluid immersion requirements.
  - Shell length increased by .600 inch.
  - Standard insert assembly components and assembly procedures can be used for retaining the socket contact insert assembly in the modified shell.

- e. The increase in weight of a connector assembly for a 22-21 arrangement will be .150 pounds.
- f. Only one interfacial seal will be required since the molded rubber insert will provide a circumferential seal at shell I.D.
- g. Connectors with interfacial inserts will mate with standard connectors. The interfacial inserts manufactured in this study are designed for use in modified MIL-C-26482 series II connectors.

## 2.4 Concept Comments

2.4.1 The following comments are applicable to the interfacial insert with regard to anomalies and advantages listed in the Disclosure of Invention Attachment 2 of subject contract.

2.4.1.1 Defects such as damaged seals, recessed contacts and bent or broken pins present a serious problem for any connector assembled on final system hardware.

2.4.1.1.1 The interfacial insert can do much to eliminate the cause of damaged seals and bent or broken pins due to improper assembly, since contacts are molded in place and seals are assembled over pins. The problem of damaged seals and bent or broken pins after assembly is eliminated since interfacial inserts are easily removed and replaced with new interfacial inserts without removing connectors from harnesses or housings.

2.4.1.1.2 The interfacial insert will have no effect on the problem of recessed contacts since this is strictly a problem of improper assembly.

2.4.2 The advantages as stated in the disclosure of invention will be discussed in the order in which they appear.

2.4.2.1 Increased Reliability - It is true that problems caused by the above anomalies will be reduced to a minimum with the incorporation of the interfacial insert. However, the amount of contacts required to make an electrical connector will increase by 50%, mating connections will increase by 100%, one additional assembly will be required and a sealing joint will also be added. It is felt that without the benefit of final design testing and in the absence of field use data the reliability will be similar to the connector being replaced.

2.4.2.2 Simplified Maintenance - The prime objective is to simplify maintenance by minimizing the effect of the stated anomalies. This interfacial insert will provide an easy means of replacing damaged interfacial seals or contacts, without costly reworking of assembled and sealed systems.

- 2.4.2.3     Interfacial Seal Damage due to Contact Installation or Removal Tools - The interfacial insert provides a molded-in contact, eliminating the problems of contact insertion and removal.
- 2.4.2.4     Elimination of Damage to Interfacial Seals due to Bonding to a Rigid Surface - It is not felt that this is an advantage since most interfacial seals are bonded to rigid surfaces. It should be noted that, in the design proposed by Bendix, the interfacial seal is bonded to a resilient member. It is the elimination of assembly and disassembly in the interfacial seal area that will reduce seal damage.
- 2.4.2.5     Reduction of Recessed Pins - It is not felt that this problem is reduced since the design only replaces insertable pins with insertable sockets.
- 2.4.2.6     Reduction of Pin Bending by Using a Harder Pin Material - Due to the molding operation, contact materials will be of the same hardness for size 20 and size 22 contacts, and size 16 contacts will be somewhat softer using the molded in contacts as opposed to crimp contacts. Note that because the pin contacts are molded in a resilient insert the bending moment due to probe type damage will be reduced.
- 2.4.2.7     Ease of Replacing Insert - Defective insert assemblies are easily removed by pulling gently with a long-nosed pliers or similar tool on various contacts in turn, so as to ease the assembly out of its shell. A new insert assembly can readily be assembled by visually aligning insert with pattern and setting in place by use of a simple hand tool.
- 2.4.2.8     Stocking of Only Socket Contacts and Tools - The interfacial insert will eliminate the need to stock individual pin contacts. However, interfacial inserts will have to be stocked. Tools for contact insertion and removal are usually common for both pins and sockets so this is only a very minor problem. Also, the interfacial insert requires its own assembly tool.
- 2.4.2.9     Only Socket Contacts on Harnesses to be Exposed - The interfacial insert will eliminate the need to have exposed pins on harnesses during harness manufacturing.
- 2.4.2.10     Crimp Tools - Usually only one crimp tool is required for either pins or sockets.
- 2.4.2.11     Selective Plating to Change Wear Patterns - Rather than change platings to shift wear areas, it is suggested to use standard platings and when pin surfaces wear, replace the interfacial insert, utilizing interfacial insert as a connector saver.

2.4.2.12 Interfacial Insert Connectors-Modified Standard Connectors - While the changes to the standard shell are simple, the interfacial insert connector can not be made from standard shells. Insert assemblies symmetrical about the main key or keyways and socket contacts are standard components, and normal assembly methods can be used.

2.4.2.13 Hermetic Applications - Interfacial inserts with pins on one end and sockets on the other could be utilized on hermetic assemblies. It might be advantageous to use socket hermetic connectors and mate with standard interfacial insert assembly connectors.

2.4.3 The disadvantages as stated in the disclosure of invention will be discussed in the order in which they appear.

2.4.3.1 Increase in weight and length - The increases in weight and length for a MIL-C-26482F series II wall mount having 21 size 16 contacts will be .15 pounds and .600 inches. The effect of this increased weight against the gain of easily repairing damaged connectors must be evaluated for each application.

2.4.3.2 Cost of adding an extra interfacial seal - In the Bendix proposal only one interfacial seal is required. This is accomplished by using the resilient insert to provide a circumferential seal with the shell.

2.4.3.3 Tooling costs - Costs to implement the interfacial insert must be compared with the saving in eliminated down-time caused by damaged pins.

2.4.3.4 Increase in contact resistance - While there will be an increase in contact resistance of a mated connector, it should be noted that on the units tested the contact resistance upper control limit did not exceed the specification requirements.

## 2.5 Test Program

### 2.5.1 Samples

15, L-25274-604 Interfacial Inserts  
4, 10-483523-21S Standard Socket Inserts  
84, 10-497007-16H Standard Socket Contacts

### 2.5.2 Test Schedule

Test Schedule  
Interfacial Inserts Only  
L-25274-604

L-25274-604

[illegible]

- 2.5.3 Test purpose - The above samples and test schedule are to provide preliminary information on the suitability and possible design problems for incorporating the interfacial insert in a modified connector.
- 2.5.4 Test Description - Paragraph references are in MIL-C-26482F unless otherwise noted.
- 2.5.4.1 Insulation Resistance Par. 3.6.7.1., Par. 4.6.8.1 except samples are not wired.
- 2.5.4.2 High Potential Par. 3.6.9.1, Par. 4.6.10.1.
- 2.5.4.3 Contact Resistance - Record contact resistance after wiring two standard socket contacts (6 in. method) and mating sockets to both sides of the pin contact at .100 in from the interfacial insert body.
- 2.5.4.4 Contact Retention - Test two contacts per interfacial insert for contact retention. Apply loads to both ends of pin contact. Apply 20 pounds maximum load.
- 2.5.4.5 Insertion Forces Interfacial Insert to Simulated Shells - Record the axial force required to insert an interfacial insert into a simulated plug shell with a standard socket insert.
- 2.5.4.6 Interfacial Insert Removal - Remove interfacial insert from shell by applying small axial loads to the pin contacts.
- 2.5.4.7 Contact Retention to Destruction - Record the axial load at which the pin contact will be dislodged from the interfacial insert. Test four contacts per interfacial insert, two contacts from each side.
- 2.5.4.8 Temp Life - Par. 3.6.37 and Par. 4.6.37.
- 2.5.4.9 Insertion Forces of Interfacial Insert and Socket Insert - Record the axial force required to assemble interfacial insert into standard insert and contact assembly.
- 2.5.4.10 Thermal Shock - Par. 3.6.13 and Par. 4.6.13.
- 2.5.5 Test Results
- 2.5.5.1 Insulation Resistance - All units passed the 5000 megohm min requirement. See data sheet 1.
- 2.5.5.2 High Potential - All samples passed the high potential at 2300 VAC. See data sheet 2.
- 2.5.5.3 Contact Resistance - The combination of two mating joints and extra length did not cause millivolt drops in excess of the 45 MV drop min. See data sheet 3.

2.5.5.4 Contact Retention - Due to the resiliency of the insert material contact retention was not run to 25 pounds. The location of the contacts at a given load and the load required to obtain a given displacement was recorded. See data sheets 4 & 5. The reason for the larger displacements when applying pushout loads from the rear of the insert assembly is the interfacial seal is a softer rubber than the insert and was yielding under the pushout load. Displacement of the contact is occurring at .0055/pounds.

2.5.5.5 Contact Retention to Destruction - The average force to release the contacts was 7.96 pounds. See data sheet 8.

2.5.5.6 Temp Life - There was no damage to inserts, seals or contacts; however, the bond at the insert-seal interface did not hold up. The force to release contacts increased to 13.66 pounds average.

2.5.5.7 Temp Cycle - There was no damage to inserts, seals, contacts or bond joints. The force required to release the contacts was 7.64 pounds average.

2.5.5.8 Simulated Shell Insertion Forces - The forces to insert the interfacial insert into three different shell openings were recorded using an insert with a full complement of mating socket contacts to push the interfacial insert into position. The forces required to remove the interfacial insert by pulling on the mated insert were also recorded. See data sheet 6. The force to insert the interfacial insert into a standard insert and contact assembly was also recorded. See data sheet 7.

The results of this test indicate that a shell I.D. of 1.155 in. and an insert with a full complement of contacts installed in the shell would retain the interfacial insert in its required position after unmating an insert with a full complement of contacts.

A connector assembly utilizing an assembled insert, interfacial insert and an I.D. of 1.155 in. was mated and unmated 50 times with an insert with a full complement of contacts. The interfacial insert did not come out of the shell during the durability testing.

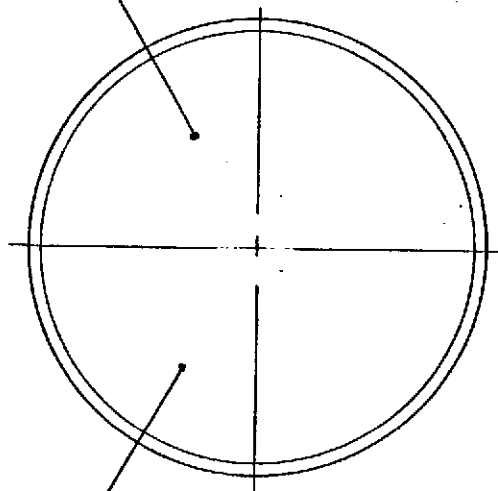
2.5.5.9 Interfacial Insert Removal - By applying small axial loads to the pin contacts of the interfacial insert, the interfacial insert was easily removed from an assembled connector.



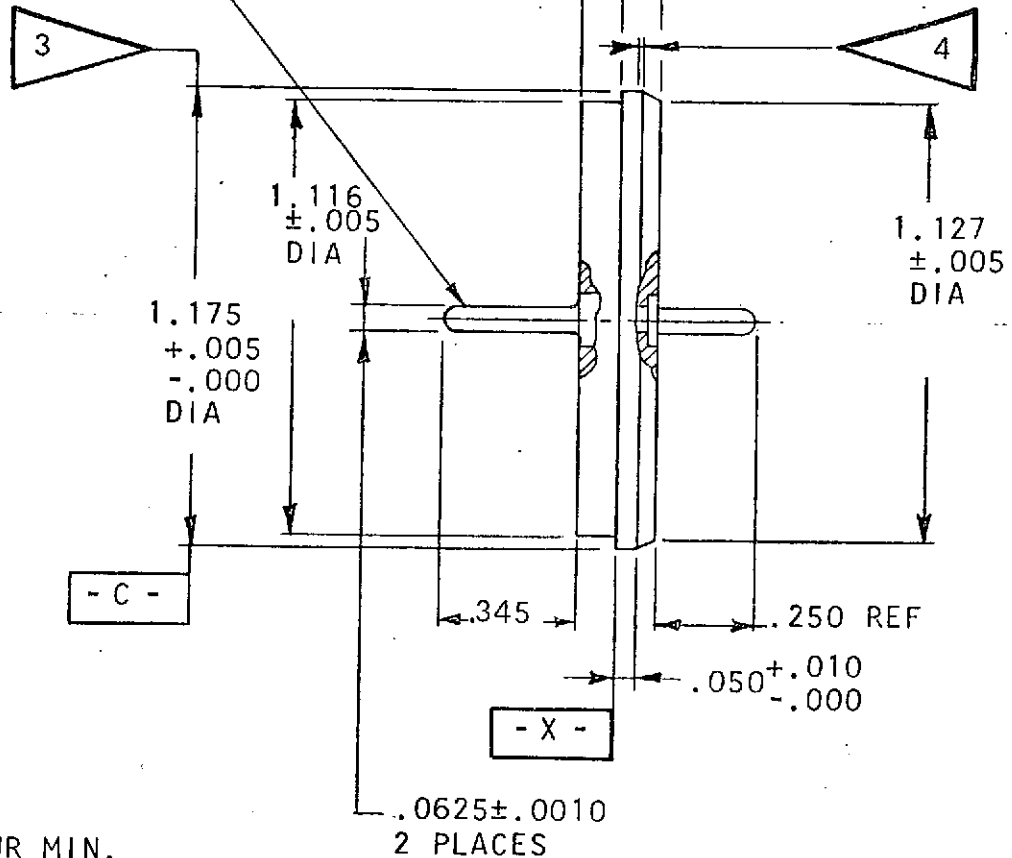
## NEXT ASSEMBLY

## REVISIONS

LTR	DESCRIPTION	DATE	APPROVED
A3	REDRAWN, WAS C SIZE ECN: CNR	2-4-75 BARTKOW	

FOR INSERT PATTERN  
SEE L-21822-21

.003

CONTACT PART NO.  
L-25274-606

- C -

- X -

1. ZZ-R-765 CLASS 2A 70 DUR MIN.

## NOTES:

⊕ C Ⓜ X .004 DIA Ⓜ

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## UNLESS OTHERWISE SPECIFIED

## SIGNATURES

## DATES

LINEAR DIMENSIONS ARE IN INCHES  
TOLERANCES: .XXX ± .010  
.XX ± .03 .X ± .1 ANGLES ± 2°  
OTHER STANDARDS PER 9-3800  
AND MIL-D-1000. NOTE REF

PREPARED

BARTKOW

2-3-75

CHECKED

MATERIAL

APPROVED

APPROVED

The Bendix Corporation  
Electrical Components Division  
Sidney, New York

INSERT & CONTACT, PIN, MOLDED,  
ELECTRICAL CONNECTOR, TYPE JTC,  
.00005 GOLD PLATED CONTACTS

MATERIAL SPEC  
MOLDING COMPOUND  
SILICONE RUBBER  
SEE NOTE 1

PROCESS SPEC  
9-3856-6  
SEE NOTE 2

SIZE

A

CODE IDENT NO.

77820

L-25274-605

SCALE NONE

WT

REF 10-25820650-()P

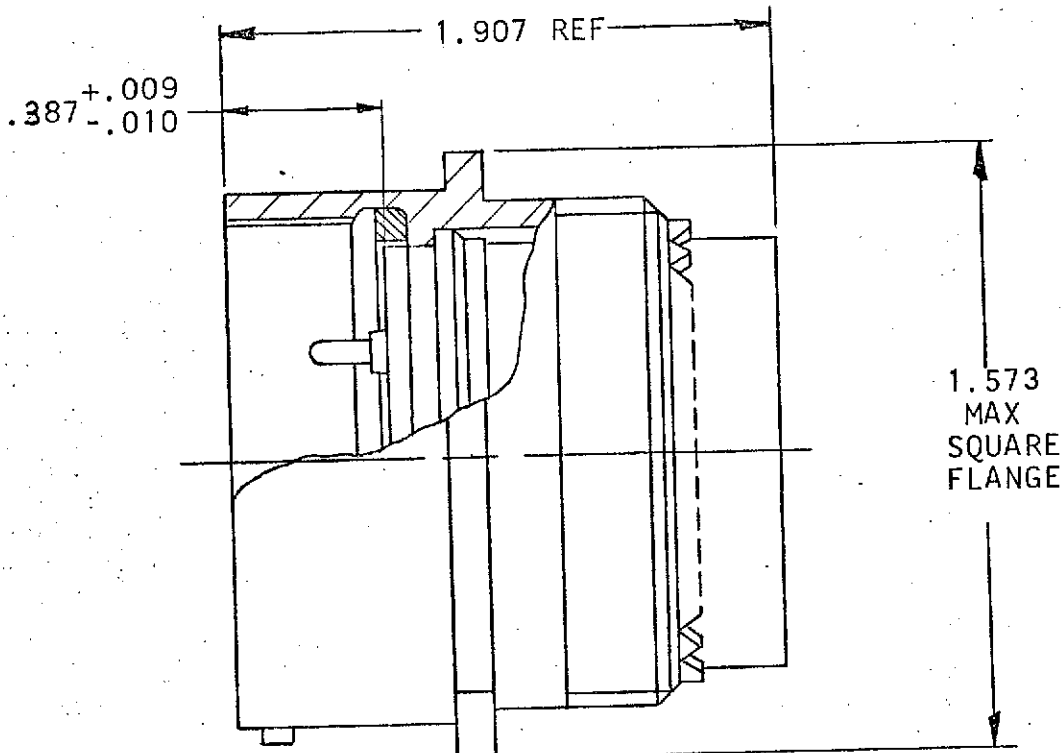
SHEET 1 OF 2

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## REVISIONS

NEXT ASSEMBLY

LTR	DESCRIPTION	DATE	APPROVED
A1			
A2	ADDED DIMENSIONS ECN: CNR	2-3-75 BARTKOW	



UNLESS OTHERWISE SPECIFIED

SIGNATURES

DATES

LINEAR DIMENSIONS ARE IN INCHES  
TOLERANCES: .XXX ± .010  
.XX ± .03 .X ± .1 ANGLES ± 2°  
OTHER STANDARDS PER 9-3800  
AND MIL-D-1000. NOTE REF ◁

PREPARED

A. HANNI

7-2-73

CHECKED

MATERIAL

APPROVED

D. MacGraw

7-9-73

APPROVED

The Bendix Corporation  
Electrical Components Division  
Sidney, New York

INTERFACIAL INSERT,  
ASSEMBLED IN MODIFIED  
MIL-C-26482 CONNECTOR, SHELL SIZE 22

U

MATERIAL SPEC

PROCESS SPEC

SIZE

A

CODE IDENT NO.

77820

L-25274-602

SCALE

NONE

WT

REF

SHEET

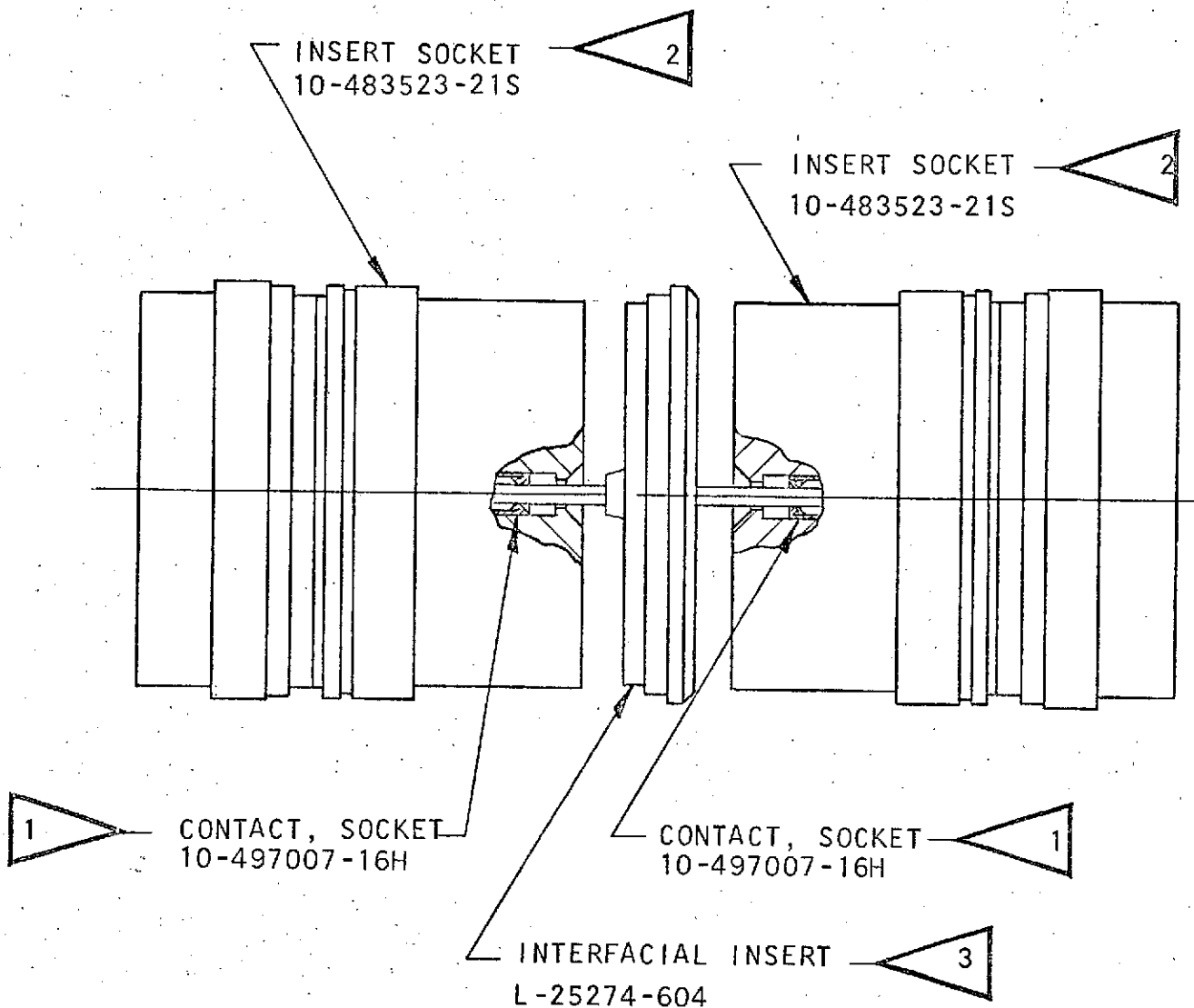
1 OF 2



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# REVISIONS

NEXT ASSEMBLY	LTR	DESCRIPTION	DATE	APPROVED
	A <sub>1</sub>			
	A <sub>2</sub>	REVISED ECN: CNR	2-3-75 BARTKOW	



2. INSERTS SHIPPED UNASSEMBLED.
1. CONTACTS SHIPPED UNASSEMBLED.

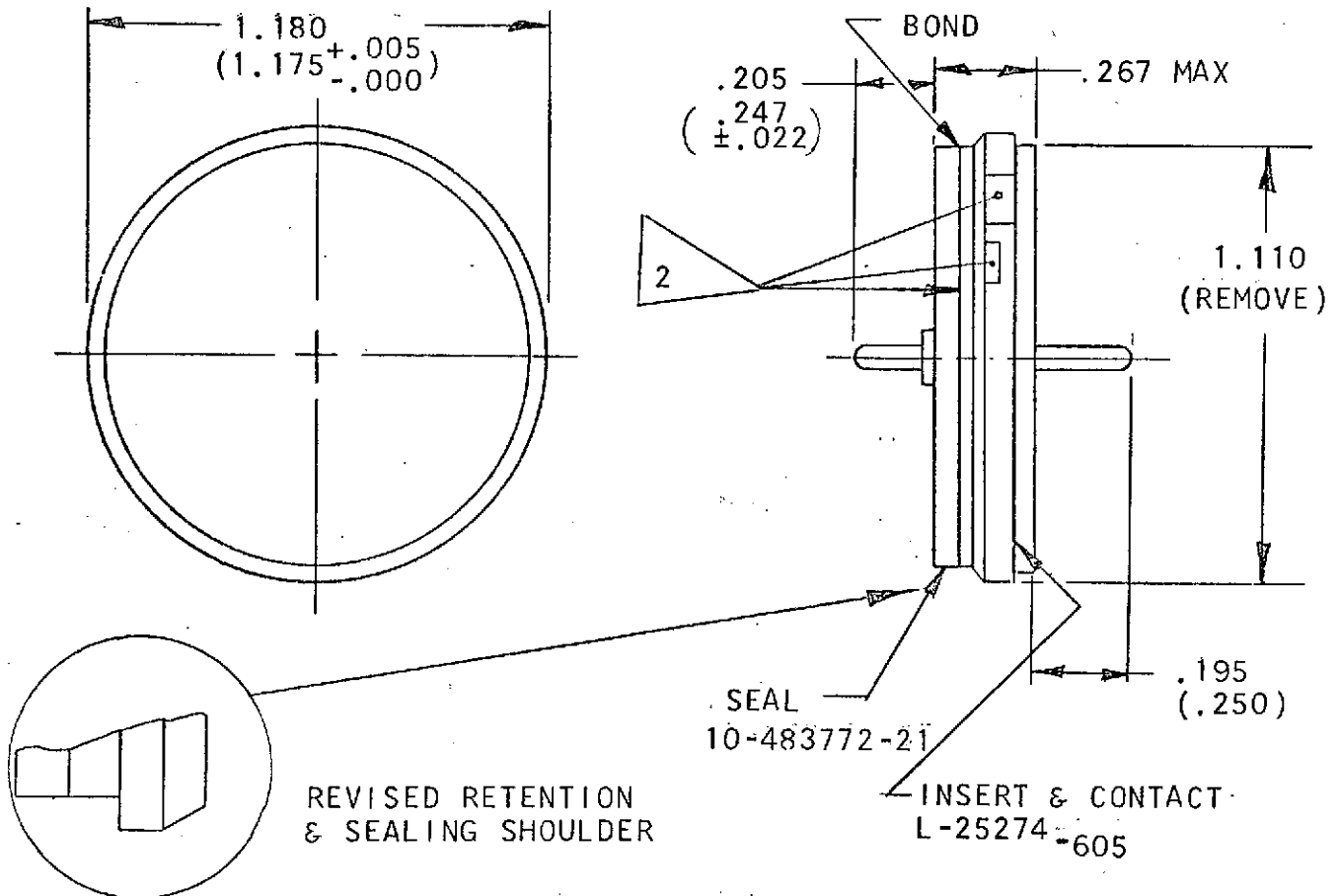
## NOTES:

UNLESS OTHERWISE SPECIFIED LINEAR DIMENSIONS ARE IN INCHES TOLERANCES: .XXX ± .010 .XX ± .03 .X ± .1 ANGLES ± 2° OTHER STANDARDS PER 9-3800 AND MIL-D-1000. NOTE REF	SIGNATURES	DATES	The Bendix Corporation Electrical Components Division Sidney, New York			
	PREPARED A. HANNI	7-2-73				
MATERIAL SPEC	CHECKED		INTERFACIAL ASSEMBLY KIT			
	MATERIAL					
PROCESS SPEC	APPROVED <i>D. MacGraw</i>	7-9-73	U			
	APPROVED					
SIZE A			CODE IDENT NO. 77820		L-25274-603	
SCALE NONE			WT	REF	SHEET	

# NEXT ASSEMBLY

# REVISIONS

LTR	DESCRIPTION	DATE	APPROVED
A <sub>1</sub>			
A <sub>2</sub>	CHANGED SEAL P/N ECN; CNR		
A <sub>3</sub>	WAS L-25274-606 ECN; CNR	4-26-74	<i>MacCoy</i>
A <sub>4</sub>	Rev. Picture	6-3-74	<i>MacCoy</i>
A <sub>5</sub>	ADDED DIMENSIONS ECN; CNR	8-30-74	
A <sub>6</sub>	REVISED ECN; CNR	2-3-75 BARTKOW	

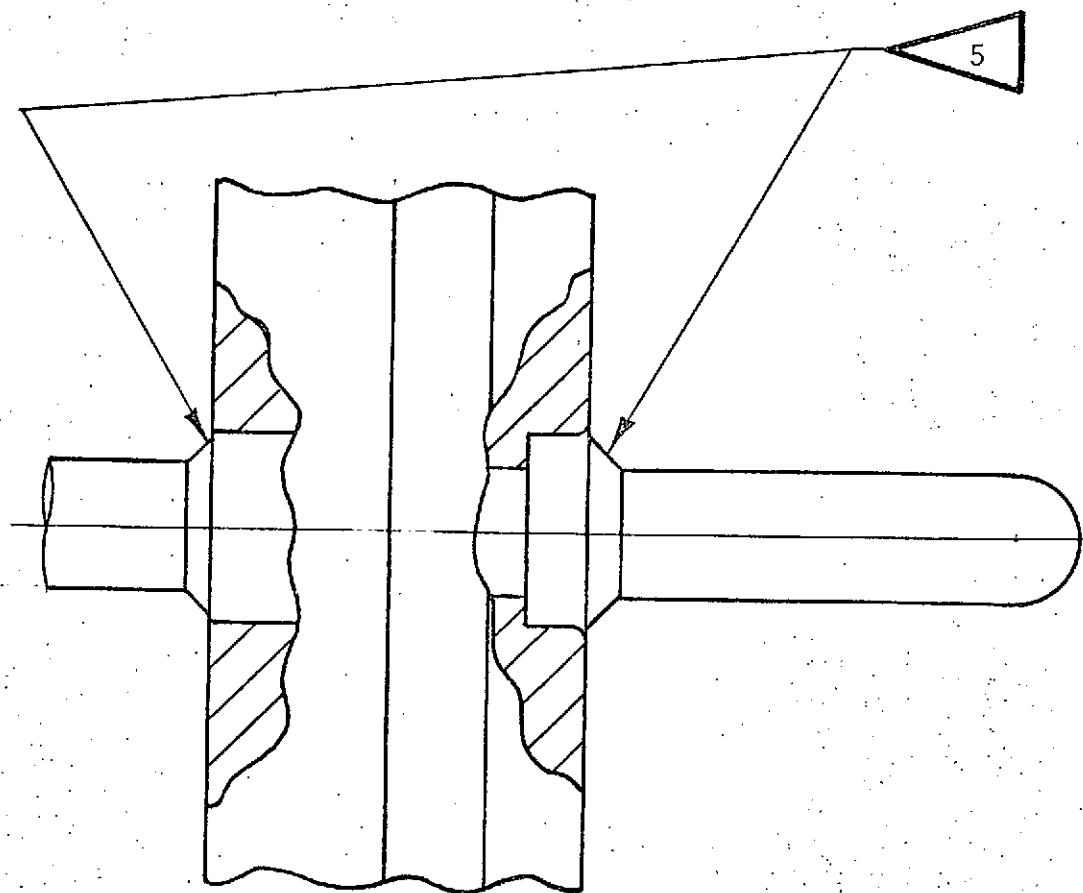


2. FULL SHOULDER GATE RELOCATED.

1.( )DIMENSIONS ARE PROPOSED CHANGES TO ACTUAL MEASURED DIMENSIONS.

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UNLESS OTHERWISE SPECIFIED		SIGNATURES		DATES		The Bendix Corporation Electrical Components Division Sidney, New York	
LINEAR DIMENSIONS ARE IN INCHES TOLERANCES: .XXX ± .010 .XX ± .03 .X ± .1 ANGLES ± 2° OTHER STANDARDS PER 9-3800 AND MIL-D-1000. NOTE REF ◁		PREPARED A. HANNI		7-2-73			
MATERIAL SPEC		MATERIAL				INTERFACIAL INSERT ASSEMBLY FOR USE ON MODIFIED MIL-C-26482 SERIES II SHELLS SIZE 22	
PROCESS SPEC		APPROVED <i>D. MacCoy</i>		7-9-73			
		APPROVED				L	
		SIZE A		CODE IDENT NO. 77820		L-25274-604	
		SCALE NONE		WT		SHEET	



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5. EITHER OR BOTH FACES OF CONTACT MAY BE DISTORTED DUE TO CLAMPING IN THE MOLD.
4. TRIM GATES .000-.005 BELOW C. GATE NOT TO EXCEED .020 IN WIDTH.
3. NO FLASH PERMISSIBLE ON INDICATED AREA OF CONTACTS.
2. USING WHITE INK PRINT PER 9-3856-6 (PROCEDURE I OR II) IDENTIFYING LETTERS AND NUMERALS IN ACCORDANCE WITH AND IN POSITION SHOWN ON L-21822-22. ON THE REAR FACE OF INSERT IT IS PERMISSIBLE TO OMIT INSERT ARRANGEMENT NUMBER AND TRADEMARK. CONTACT IDENTIFICATION ON REAR FACE MUST INCLUDE ALL LETTERS AND NUMERALS LOCATED ON VERTICAL CENTERLINE AND OUTER PERIPHERY IF IDENTIFIED ON L-21822-21. LETTER HEIGHT TO BE .047 APPROX UNLESS OTHERWISE SPECIFIED ON L-21822-21.
1. SPECIFIED GENERAL TOLERANCES:  
 CORNERS SHARP .000-.010  
 CONCENTRICITY .005 TIR (RFS)

NOTES:

REV SYM		A1		SIZE A		CODE IDENT NO. 77820		L-25274-605	
SCALE		WT		REF		SHEET		2	

[illegible]

# LABORATORY DATA SHEET

TEST <i>DIELECTRIC WITHSTANDING VOLTAGE (SEA LEVEL)</i>				DATE OF TESTS <i>8-20-74</i>		REPORT NUMBER	
TEST SPECIMEN(S) <i>QTY 15 INTERFACIAL INSERTS w/o SHELL</i>				TEMP. <i>174°F</i>		R.H. <i>41%</i>	
<i>L-25274-604</i>				SPEC. <i>MIL-C-26482F</i>		ECL <i>4.6.10.1</i>	
TEST EQUIPMENT <i>BK AUTOMATIC CONN. TESTER</i>				CAL. DATE <i>8-8-74</i>		DUE DATE <i>9-6-74</i>	
				SPEC. LIMITS			
				TEST CONDITIONS			
				<i>METHOD 301</i>			
				<i>MIL-STD-202E</i>			
<p><i>THE DIELECTRIC WITHSTANDING VOLTAGE WAS APPLIED BETWEEN EACH CONTACT AND ALL OTHER CONTACTS IN COMMON.</i></p> <p><i>THE TEST VOLTAGE WAS GRADUALLY APPLIED AT RATE OF 500 VOLTS PER SECOND, UNTIL A MAXIMUM VOLTAGE OF 2300 VAC RMS WAS ATTAINED.</i></p> <p><i>THE MAXIMUM TEST VOLTAGE WAS MAINTAINED FOR ONE MINUTE.</i></p> <p><i>NO EVIDENCE OF FLASHOVER, BREAKDOWN, OR LEAKAGE CURRENT IN EXCESS OF ONE MILLIAMPERE.</i></p>							
<p><b>ORIGINAL PAGE IS OF POOR QUALITY</b></p>							
REF.						TESTED BY <i>E. R. Bohrerka</i>	
				Electrical Components Division Sidney, New York 13838		APPROVED BY	
						WITNESSED BY	



# LABORATORY DATA SHEET

TEST <b>CONTACT RESISTANCE</b>			DATE OF TESTS <b>8-21-74</b>		REPORT NUMBER
TEST SPECIMEN (S) <b>QTY 2 WIRED 10-483523-218 INSERTS</b>			TEMP. <b>+75°F</b>	R.H. <b>36%</b>	ECL <b>50%</b>
QTY 1 L-25274-604 INTERFACIAL INSERT			LT. <b>3735</b>	SPEC. <b>MIL-C-26482E</b>	PARA. <b>4.6.5</b>
TEST EQUIPMENT		CAL. DATE	DUE DATE		SPEC. LIMITS
<b>KEPCO POWER SUPPLY</b>		<b>E1375</b>	<b>6-28-74</b>		<b>7-1-75</b>
<b>3450A H.P. MULTIMETER</b>		<b>E1822</b>	<b>4-11-74</b>		<b>10-8-74</b>
<b>9820 H.P. CALCULATOR</b>					
TEST CONDITIONS					
<b>45 MILLIVOLTS</b>					
<b>6" METHOD</b>					
<b>10 AMPERES DC</b>					
<b>MATED ASSEMBLIES</b>					

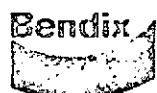
## STAT. ANALYSIS DATA IN MV

37.49  
36.97  
36.89  
36.89  
38.36  
37.43  
37.92  
37.89  
37.56  
37.04  
37.34  
36.64  
37.98  
36.57  
36.91  
36.72  
37.95  
40.00  
36.77  
37.04  
37.00

SAMPLES= 21  
MAXIMUM= 40.00  
AVERAGE= 37.36  
MINIMUM= 36.57  
STD. DEV.= .78  
3S U.C.L.= 39.70  
3S L.C.L.= 35.02

ORIGINAL PAGE IS  
OF POOR QUALITY

REP.



Electrical  
Components  
Division

3

Sidney, New York 10838

TESTED BY

APPROVED BY

WITNESSED BY

# LABORATORY DATA SHEET

TEST <b>CONTACT RETENTION</b>				DATE OF TESTS <b>8-22-74</b>		REPORT NUMBER	
TEST SPECIMEN(S) <b>QTY 15 NASA INTERFACIAL INSERTS</b>				TEMP. <b>75°F</b>		R.H. <b>38%</b>	
<b>L-25274-604</b>				SPEC. <b>3735</b>		ECL <b>PARA.</b>	
TEST EQUIPMENT <b>PG1050</b>		CAL. DATE <b>3-28-74</b>		DUE DATE <b>10-4-74</b>		SPEC. LIMITS	
<b>INSTRON TTC. W/EXTENSOMETER</b>		<b>4-26-74</b>		<b>5-17-76</b>		TEST CONDITIONS <b>LOAD APPLIED TO FRONT OF DOUBLE PIN CONTACT (INTERFACIAL SEAL SIDE)</b>	
<b>INSTRON CALIB. WEIGHTS PG1450</b>							
SAMPLE #	LOAD @	DEFL. @		SAMPLE #	LOAD @	DEFL. @	
CONTACT #	.015 DEFL.	4 LBS. LOAD		CONTACT #	.015" DEFL.	4 LBS. LOAD	
	(POUNDS)	(INCHES)			(POUNDS)	(INCHES)	
1A	3.8	.016		9W	3.3	.019	
1C	3.3	.020		9S	3.1	.020	
2P	2.7	.024		10S	2.5	.025	
2B	2.8	.025		10T	2.6	.024	
3D	2.8	.021		11T	3.1	.020	
3E	2.6	.023		11U	2.7	.022	
4X	3.0	.022		12W	3.6	.017	
4S	2.9	.023		12V	3.0	.022	
5A	2.4	.027		13J	2.4	.025	
5B	2.2	.027		13H	3.0	.021	
6C	2.6	.025		14R	2.6	.023	
6D	2.5	.026		14S	2.9	.022	
7H	2.6	—		15C	2.2	.025	
7J	3.0	—		15D	2.4	.024	
8N	2.4	.025		Average	2.80	.0227	
8M	2.7	.022					
REF.				TESTED BY <b>E. R. Bohner</b>			
				APPROVED BY			
				WITNESSED BY			



**Electrical Components Division**

Sidney, New York 13838

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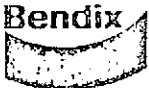
# LABORATORY DATA SHEET

TEST <b>CONTACT RETENTION</b>				DATE OF TESTS <b>8-22-74</b>		REPORT NUMBER	
TEST SPECIMEN(S) <b>QTY. 15 NASA INTERFACIAL INSERTS</b>				TEMP. <b>+75°F</b>		R.H. <b>38%</b>	
L-25274-604				SPEC.		ECL EPL PARA.	
TEST EQUIPMENT <b>PG1050 INSTRON TTC 14/EXTENSOMETER</b>		CAL. DATE <b>3-28-74</b>		DUE DATE <b>10-4-74</b>		SPEC. LIMITS	
INSTRON CALIB WEIGHTS PG1450		4-26-74		5-17-76		TEST CONDITIONS <b>LOAD APPLIED TO REAR OF DOUBLE PIN CONTACT.</b>	

SAMPLE#	LOAD (LBS)	DEFL. (IN)		SAMPLE#	LOAD (LBS)	DEFL. (IN)	
CONTACT#	.015" DEFL (POUNDS)	4 HRS LOAD (INCHES)		CONTACT#	.015" DEFL (POUNDS)	4 HRS LOAD (INCHES)	
1 N	2.7	.039		9 X	1.8	.029	
1 L	1.0	.042		9 U	1.9	.028	
2 U	1.3	.047		10 P	1.9	.029	
2 V	1.7	.045		10 R	1.9	.031	
3 L	0.7	.039		11 C	1.5	.037	
3 K	1.4	.033		11 D	1.6	.035	
4 G	1.2	.041		12 X	1.7	.032	
4 H	1.7	.041		12 S	1.8	.031	
5 N	2.0	.026		13 E	1.8	.033	
5 M	2.2	.027		13 D	1.8	.034	
6 U	1.6	.032		14 S	1.3	.037	
6 H	1.7	.037		14 T	2.1	.030	
7 V	1.5	.032		15 K	1.2	.044	
7 K	1.0	.042		15 J	1.6	.040	
8 A	1.1	.039		Average	1.61	.0354	
8 B	1.5	.039					

REF.	 <b>Bendix</b> <b>Electrical Components Division</b> <b>5</b> Sydney, New York 10918	TESTED BY <i>J. R. Bohrer</i> APPROVED BY WITNESSED BY
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